INSTABILITIES IN A SHOCK INTERACTION WITH A PERTURBED CURTAIN OF PARTICLES

RICARDO GONZALEZ IZARD^{1,2}, SUMANTH REDDY LINGAMPALLY³, PATRICK WAYNE³, GUSTAAF JACOBS² & PETER VOROBIEFF³

¹Department of Fluid Mechanics, The University of Valladolid, Valladolid, Spain. ²Department of Aerospace Engineering, San Diego State University, San Diego, CA, USA. ³Department of Mechanical Engineering, The University of New Mexico, Albuquerque, NM, USA.

ABSTRACT

We present a two-dimensional computational study of a shock interaction with a particle-seeded curtain where particles initially comprise 4% by volume, and the rest is air. If the initial depth of the curtain in the streamwise direction is variable, numerical results predict vortex formation in both the gas phase and the dispersed phase after the shock-curtain interaction. The phenomenon is distinct from baroclinic (Richtmyer–Meshkov) instability observed on gaseous density interfaces and is caused by the changes in the particle phase number density distribution and related interphase velocity changes. *Keywords: baroclinicity, CFD, particle-laden flow, Richtmyer–Meshkov instability, WENO-Z*

1 INTRODUCTION

Hydrodynamic instabilities at the interface of two materials of different densities are a critical issue in high energy density physics (HEDP). The Rayleigh–Taylor instability (RTI) occurs when a fluid accelerates another fluid of high density [1,2]. The RTI is ubiquitous in HEDP phenomena, such as high Mach number shocks and jets, radiative blast waves and radioactively driven molecular clouds, gamma-ray bursts and accreting black holes. Due to the importance in physics mentioned above, there have been many studies related to RTI from both physical and numerical simulation points of view in the literature.

It was shown [3] that when two immiscible fluids of different densities are accelerated in a direction perpendicular to their interface, this interface is stable or unstable according to whether the acceleration is directed from the heavier to the lighter fluid or vice versa. The case with gravity (with acceleration g pointing vertically downwards) is equivalent to the two fluids being accelerated vertically upwards with the same acceleration g.

A wide variety of fluid motions can be generated following the interaction of a shock wave with an interface separating two fluids of different properties. Any perturbation initially present on the interface will, in most cases, be amplified following the refraction of the shock. This class of problems is generally referred to as the Richtmyer–Meshkov instability (RMI). The basic mechanism for the amplification of perturbations at the interface is baroclinic vorticity generation resulting from misalignment of the pressure gradient of the shock and the local density gradient across the interface. The growth of perturbations soon enters into a nonlinear regime with the appearance of bubbles of light fluid rising into heavy fluid and spikes of heavy fluid falling into the light fluid. As the interface between the two fluids becomes more distorted, secondary instabilities, such as the Kelvin–Helmholtz shearing instability, develop, and a region of turbulence and mixing (usually referred to as mixing zone) ultimately forms.

The first analytical study on impulsive acceleration of an interface by a shock wave was performed by Markstein [4], who investigated the interaction of a shock wave with a density interface (flame front). His analytical result was the same as Taylor's result for constant acceleration [2]. But the first real treatment of the impulsive acceleration of the interface by

a shock wave was given by Richtmyer [3]. He studied the growth of a sinusoidally perturbed interface in the linear regime following the shock wave impingement. In developing his impulsive model, he assumed that the shock wave is not strong enough to cause perturbation velocities comparable to the speed of sound. Ten years later, Meshkov [5] confirmed Richtmyer's prediction experimentally.

In studies of RMI an interface between two species with different density is usually considered. In detonations and explosion, a discrete phase of particles (also called dispersed phase) is often a part of the physical setting. In dispersed phase flows, particles that are materially not connected to each other form a separate phase. These include gas-particle and liquid-particle flows in which the particles constitute the dispersed phase. Few studies report on the instabilities that occur on the interface of a curtain of particles interacting with a shock. Balakrishnan *et al.* [6] investigate the flow field subsequent to the detonation of a spherical charge of TNT with an ambient distribution of dilute aluminum particles. Jacobs *et al.* [7] have studied the interaction between blast waves and clouds of particles.

In this paper, we study the interaction between a moving, normal shock and a perturbed curtain of particles. We show that the curtain exhibits instabilities that are very similar to RMI and RTI, however, as earlier research [8, 9] indicates, these instabilities are non-baroclinic, but rather driven by interphase velocities and changes in number density in the particle phase. Here and further on, 'interphase velocity' refers to a velocity difference between the gaseous and particle phases.

In the next section, we present the physical and numerical model. We then discuss the qualitative features of a shock interaction with a sinusoidal perturbed curtain of particles. The growth rates are compared with RMI theory in the next section. Conclusions are reserved for the last section.

2 GOVERNING EQUATIONS AND NUMERICAL MODEL

We model the particle-laden flow with the particle-source-in-cell (PSIC) method [10]. The conservation equations are solved for the carrier flow in the Eulerian frame, while particles are traced in the Lagrangian frame. In the following, we shall denote the subscript p for the particle variables and f for the gas variables at the particle position.

2.1 Gas Phase

The governing equations for the carrier flow are the inviscid two-dimensional Euler equations in Cartesian coordinates given by:

$$\vec{Q}_i + \vec{F}_x^a + \vec{G}_y^a = \vec{S} \tag{1}$$

where

$$\vec{Q} = (\rho, \rho u, \rho v, \rho e)^{T},$$

$$\vec{F}_{x}^{a} = (\rho u, \rho u^{2} + P, \rho u v, (\rho E + P)u)^{T},$$

$$\vec{G}_{y}^{a} = (\rho v, \rho u v, \rho v^{2} + P, (\rho E + P)v)^{T},$$
(2)

are the advective fluxes, and the pressure is determined by

$$P = (\gamma - 1) \left(\rho E - \frac{1}{2} \rho (u^2 + v^2) \right)$$
(3)

The equation of state closes the system

$$T = \frac{T_{ref}}{U_{ref}^2} \frac{\gamma P M^2}{\rho}$$
(4)

Here $M = U_{ref} / \sqrt{\gamma RT_{ref}}$ is a reference Mach number determined with the reference velocity U_{ref} and reference temperature T_{ref} . The source term \vec{S} accounts for the effect of the particles on the carrier gas and will be discussed in more detail below.

2.2 Dispersed Phase

The particle phase is solved by tracking the so-called point particles individually using equations for an empirically corrected Stokes flow around a spherical particle. The kinematic equation describing the particle position x_n is given by

$$\frac{d\vec{x}}{dt} = \vec{v}_p \tag{5}$$

The particle acceleration is governed by Newton's second law forced by the drag on the particle. With particles assumed spherical, we take the drag as a combination of the Stokes drag with corrections for high Reynolds and Mach number and the pressure drag leading to the following equations governing the particle velocity:

$$\frac{d\vec{v}}{dt} = f_1 \left(\frac{\vec{v}_f - \vec{v}_p}{\tau_p} \right) - \frac{1}{\rho_p} \nabla P \mid_f$$

Here, \vec{v}_f is the velocity of the gas at the particle position and ρ_p is the particle density. The first term on the right hand side describes the particle acceleration resulting from the velocity difference between the particle and the carrier gas flow. f_I is an empirical correction factor [11] that yields an accurate determination within 10% of measured particle acceleration for higher relative particle Reynolds numbers up to $R_f = 10,000$ and relative particle Mach numbers up to $Mf = |\vec{v}_f| / \sqrt{T_f} = 1.2$:

$$f_1 = \frac{3}{4} (24 + 0.38R_f + 4\sqrt{R_f}) \left(1 + \exp\left[\frac{-0.43}{M_f^{4.67}}\right] \right)$$
(7)

The second term is the particle acceleration induced by the pressure gradient in the carrier flow at the particle position. The particle time constant is $\tau_p = R d_p^2 \rho_p / 18$, where d_p is the particle diameter. This time constant is a measure for the reaction time of the particle to the changes in the carrier gas. R= UL/v is the Reynolds number of the carrier gas flow, where L is the reference length and is the dynamic viscosity. Because R is large, we do not model viscous effects in the governing Euler equations for the gas flow. The particle temperature is mostly affected by convection. From the first law of thermodynamics and Fourier law for heat transfer, the equation for temperature is derived as

$$\frac{dT_p}{dt} = \frac{1}{3} \frac{Nu}{\Pr} \left(\frac{T_f - T_p}{\tau_p} \right)$$
(8)

where Pr = 1.4 is the Prandtl number, taken at its typical value for air in this paper. $Nu = 2 + \sqrt{R_r P_r^{0.33}}$ is the Nusselt number corrected for high Reynolds number.

61

2.3 Source term S

Each particle generates momentum and energy that affect the carrier flow. The volume averaged summation of all these contributions gives a continuum source contribution on the momentum and energy equation in eqn 1 as

$$S_{m}(x) = \sum_{i=1}^{N_{p}} \vec{K}(x_{p}, x) W_{m}$$

$$S_{e}(x) = \sum_{i=1}^{N_{p}} \vec{K}(x_{p}, x) W_{m} \vec{v}_{p} + W_{e}$$
(9)

where $\vec{K}(x, y) = \vec{K}(|x - y|)/V$ is a normalized distribution function that distributes the influence of each particle onto the carrier flow. W_m and W_e are weight functions describing the momentum and energy contribution of one particle and are

$$W_{m} = m_{p} f_{1} \frac{(T - T_{p})}{\tau_{p}}$$

$$W_{e} = m_{p} \frac{Nu}{3 \operatorname{Pr} \tau_{p}} (T - T_{p})$$
(10)

 m_p is the mass of one spherical particle, which can be derived from T_p and ρ_p . N_p is the total number of particles in a finite volume V.

2.4 Numerical Model

The Langrangian algorithm traces each particle's position, velocity and temperature over time. The carrier phase properties such as velocities and temperature required at each particle's position are obtained via interpolation from the grid points that surround the particle to the position of the particle. To account for the back-way coupling effect of particles on the carrier phase, the momentum and energy of the particle phase forcing is averaged and coupled to the carrier phase equations through a source term. Crowe *et al.* [10] first proposed this Particle-Source-in-Cell (PSIC) model for gas-droplet flows. The PSIC method is a particle-mesh type algorithm where the carrier phase is resolved in a static mesh, while the particle dynamics are traced along their path in the Lagrangian frame.

Here we use the high-resolution PSIC algorithm developed by Jacobs and Don and described in [7]. The Jacobs algorithm is a high-order, Eulerian–Lagrangian algorithm based on the weighted essentially non-oscillatory (WENO) finite difference method on a structured grid for the computation of the gas phase. Higher-order ENO interpolation between the Eulerian gas PDEs and the Lagrangian particle ODES ensure that the high-order nature of the WENO scheme is preserved in the EL-WENO method. The high-order resolution scheme captures shocks sharply without diffusing essential interactions between shocks, turbulence and particles.

3 PROBLEM SETUP

We simulate the interaction of a perturbed curtain with a 4% volume fraction loading of 135,000 particles with a moving planar shock as schematically summarized in Fig. 1. The particle response time and density are $\tau_p = 1.7845 \times 10^3$ and $\rho_p = 1200$, respectively.



Initial Shock Wave location

Figure 1: Schematic of the initial condition of a planar normal shock interaction with a perturbed curtain of particles.

Particles are initially evenly distributed in a perturbed curtain stretching a width of 0.2941 (in x-direction) and a height of 0.2 (in y-direction) with 150 x 900 particles. The function of the initial perturbation follows a cosine profile allocating the particles uniformly. The x and y locations of each particle are assigned as follows:

$$x_{p} = x_{0} + k \cdot \delta x_{p} + \omega_{0} \cdot \cos\left(\frac{2\pi}{T_{cloud}}\right) \cdot b \cdot y_{loc}$$
(11)

$$y_p = -\frac{b}{2} + j \cdot \delta y_p \tag{12}$$

Here x_p and y_p are the position of the particle in x and y directions, respectively, x_0 is the initial location of the curtain, $\delta x_p = \frac{a}{n_{partx}}$ is the distance between particles in x-direction, where a is the width of the curtain and n_{part-x} is the number of particles in x-direction, $\delta y_p = \frac{b}{n_{party}}$ is the distance between particles in y-direction where b is the height of the curtain and n_{part-x} is the number of particles in the curtain and n_{part-y} is the number of particles in y-direction. The variable k is a counter for each row of particles ranging from 1 to n_{part-x} while j is a counter for each column of particles ranging from 1 to n_{part-x} while j is a counter for a particle is the wavelength of the perturbation. $\delta y_p = \frac{b}{n_{party}}$ also identifies the coordinate of a particle row in y direction.

For simulation of the gas phase we use a fifth-order WENO scheme with a mesh of 800 by 200 uniformly spaced grid points in the *x* and *y* directions, respectively.

The computational domain spans an area of -0.5 to 6.5 in the x-direction, which enables the tracing of relevant instabilities in the carrier and dispersed phase. The computational domain's height ranges from -0.1 to 0.1 in the y-direction. A uniform inflow boundary condition is specified at x = -0.15 and a periodic boundary condition is specified on the spanwise boundaries of the domain. The right boundary of the domain specifies outflow boundary conditions, as shown in Fig. 2. For calculation of the viscous forcing on the particles the Reynolds number is set to $R_f = 9.6503 \cdot 10^5$. The shock Mach number is M = 2.8.



Figure 2: Boundary conditions include periodic boundaries in spanwise directions and inflow/ outflow boundary conditions. A uniform mesh with 800 x 200 points is used with a fifth order WENO finite difference scheme.

Tables 1 and 2 summarize the parameters and initial conditions of the particle and carrierphase respectively. The initial configuration of the simulation is depicted in Fig. 3 that shows the initial density of the carrier and the initial position of the particles in the cloud. Throughout this paper the color coding of the particles (Fig. 3) is used to identify the initial location of columns of particles in the curtain.

Name	Symbol	Value
Particle Response Time	T	$1.7845 \cdot 10^3$
Density of the particles	ρ_p^P	1200
Reynolds Number	R_{f}	$9.6503 \cdot 10^5$
Cloud Width	Å	0.298
Cloud Height	В	0.2
Volume fraction	V	0.04
Initial Cloud amplitude	ω_o	0.1
Period	T_{cloud}	1.0
Start location of the cloud	x_{o}	0.0
Particle distance in x-direction	δx_p	1.96066 • 10 ⁻³
Particle distance in y-direction	δy_p	2.222 • 10 ⁻⁴

Table 1: Summary of the parameters and initial condition of the particle phase.

Table 2: Summary of	of the parameters	and initial	condition of	f the carrier phase.

Fluid 1	Value	Fluid 2	Value
<i>u</i> ₁	8.824	u_2	0
v_1	0	v_2	0
ρ_1	3.66	ρ_2	1
E_1	33.0775	E_2	2.5
$\dot{M_1}$	2.78	$\dot{M_2}$	0



Figure 3: Contours of the initial density field and color-coded particle curtain location.

4 PARTICLE CURTAIN INSTABILITIES

While there are no baroclinic effects in the dispersed phase, the growth of the initial perturbation in the curtain's geometry is very similar to that of a baroclinically driven instability. The mechanism is of course very different and we will show is closely connected to changes in the local number density of the dispersed phase and related changes in the interphase velocity between the particles and the fluid. If the number density is higher, the carrier gas flow is decelerated by the denser particle phase. A lowered carrier phase velocity reduces the forcing of the particle motion by the gas phase and hence leads to a reduction in the particle phase velocity.

When the shock interacts with the (left) farthest upstream particles at an early time of t = 1.1, the middle part of the curtain is compressed, as observed in the number density $\Phi = \sum_{i}^{N_p} (K(x - x_{p,i}))$ contours in Fig. 4. The middle part of the curtain remains more



Figure 4: Density contours with color code particle locations (top), particle phase's number density Φ , contours (middle) and particle phase's averaged velocity, \tilde{U} , contours (bottom) at t = 1.1.



Figure 5: Particle phase number density F contours at t=2.2.

compressed as compared to the side parts throughout the flow development (e.g. at t = 2.2 and t = 3.3 in Figs. 5 and 6, respectively) because of this earlier interaction with the accelerated flow behind the shock. The resulting lower particle phase velocity in streamwise direction, $u = \frac{1}{\Phi} \sum_{i}^{N_p} (K(x - x_{p,i})) u_{p,i}$, in the center as compared to the sides (Fig. 4) leads to growth of the initial perturbation.

The difference in velocity magnitude in the middle and the side creates a vortex pair in the gas phase as observed by the vorticity, $\Omega = \frac{\partial v}{\partial x} - \frac{\partial u}{\partial y}$ contours in Fig. 7. The vortex pair transports particles from the sides to the middle at the downstream location of the curtain. The positive particle velocity in y-direction, \vec{v} and corresponding symmetric negative velocity on the bottom (Fig. 8), leads to an increase in the number density along the centerline (Fig. 6).

In the upstream part of the curtain the \vec{v} velocity is positive and negative on the top and bottom half of the domain, respectively. Particles are hence transported towards the outer boundaries of the domain. Because we have specified a periodic boundary condition in spanwise direction, the particles that leave the domain through the bottom boundary enter through



Figure 6: Zoomed-in particle phase number density Φ contours at t=3.3. The highlighted circle shows particles that originate from the bottom boundary and enter through the top boundary via the periodic boundary condition.



Figure 7: Vorticity contours in the gas phase at t = 2.2.



Figure 8: Particle phase velocity, V, contours at t = 2.2.



Figure 9: Particle phase velocity, V, contours at t = 3.3.

the top boundary and vice versa at later times (t > 3), as highlighted in Fig. 6. The number density trends are directly connected to the structures in the velocity contours in Fig. 9. We observe that the centerline velocity is lower in areas where the particle number density is high. The periodic particle streaks at the sides of the domain redirect the flow and create a visible change in the velocity at those locations.

5 GROWTH OF INSTABILITY

By assuming a potential flow, one can perform analysis of the acceleration driven normal instability between two fluids of different densities. Richtmyer [12] found that the growth of the perturbed interface is linear according to

$$\frac{d\omega}{dt} = k \left[u \right] A \omega_o \tag{13}$$

where ω_o is the initial amplitude, $[u] = \dot{u}_p/dt$ is the initial change in the velocity of the interface imparted by the shock wave, $A = \frac{\rho_2 - \rho_1}{\rho_1 + \rho_2}$ is the Atwood number and $k = \frac{2\pi}{\lambda}$ is the wave number of the initial perturbation.

The amplitude ω grows with the time. Both light-heavy (A > 0) and heavy-light (A < 0) cases are unstable. For A = 0 and A >0 the initial amplitude is monotonically increasing. For A <0, the amplitude is initially zero, then reverses sign and grows according to equation 13.

Richtmyer linearized the compressible computations using finite difference techniques and accounted for an initial compression of the interface using post-shock values A and u proposed by Richtmyer was

$$\frac{d\omega}{dt} = k\left[u\right]A'\omega_0' \tag{14}$$

To apply Richtmyer's formulas to a particle-laden flow, we determine the post-shock Atwood number based on density of the mixture, p_{mix} , as

$$A' = \frac{\rho_{mix} - \rho_1}{\rho_1 + \rho_{mix}} \tag{15}$$

The mixture density is determined as

$$\rho_{mix} = \rho_p \mathbf{V} + \rho_1 (1 - \mathbf{V}) \tag{16}$$

where the volume concentration of the particles is V=0.04 and p_1 is the density of the gas before the shock wave.

Using the computations of the shock-curtain interaction, we compare the computational growth rate against this theoretical growth rate for three different values of particle density, ρ_{part} =1200, 2400 and 8800. Note that if the density of the particle is changed, then the particle's diameter changes also according to

$$d_p = \sqrt{\frac{18\tau_p}{\operatorname{Re}_f \rho_p}} \tag{17}$$

where d_p is the diameter of the particle. τ_p is the particle response time and R_f is the Reynolds Number around the particle. For a fixed number of particles the volume concentration reduces with an increased particle density and a reduced particle diameter. In Table 3, we summarize the parameters for the three cases considered.

In Fig. 10, growth of the amplitude of the curtain's perturbation is plotted in time. All cases show an initial compression of the curtain followed by a linear growth. The compression of is smallest for $p_p = 1200$ and increases with increasing particle density as predicted by theory.

The growth rates for the three cases compare remarkably well with theory, considering that the instability in the particle phase is non-baroclinic. The growth rates are m=0.3828, 0.3357 and 0.2991 for $p_p=1200, 2400$ and 8800, respectively. Comparison of these growth rates

(Table 3) to the theoretical growth rates of $\frac{d\omega}{dt}$ =0.4527, 0.3766 and 0.2381, respectively, shows that the theory and computation are within at most 15% The deviation is a largest for the lowest particle density that was considered here.

instability.				
^{p}p	1200	2400	8800	
V	0.04	0.0282	0.0147	
ρl	3.66	3.66	3.66	
ρ_{mix}	51.5136	65.7	133.52	
A	0.86733	0.8961	0.9464	
[u]	0.1658	0.1338	0.08	
$\frac{d\omega}{dt}$	0.4527	0.3766	0.2381	

Table 3: Parameters and results of a parametric study of the effect of particle densities of $p_p=1200$, 2400 and 8800 on the growth rate, dw/dt, of a shock acceleration driven instability.



Figure 10: Amplitude of perturbation, u in the curtain plotted versus time, t, for three different particle densities, p_p compared against the linear theoretical growth according to Eq. 14

6 CONCLUSIONS AND FUTURE DEVELOPMENTS

A study is conducted on the growth of a perturbation in a curtain of particles when a planar shock moves through it. Two-dimensional computations show that the perturbation of the particle curtain grows much like a baroclinically driven instability in a gas phase. However, the growth mechanism is different: because of an increased particle number density at the farthest upstream part of the perturbation, an interphase velocity difference develops in spanwise direction. This in turn yields a spanwise velocity difference in the particle phase that drives the growth of the perturbation.

An important finding is that the growth rate of this instability initially follows the theoretically linear baroclinic growth. In a parametric study on the effect of the particle density, we found an agreement between the theoretical equation and the simulation within 15%. This observation is also consistent with recent experimental results [9].

Our current efforts focus on a parametric study with different dependencies to further explore instability growth in the particle phase. We are also working on an experiment based on this simulation. We expect to find similarities, not only in the behavior of the instability, but also in the growth rate.

ACKNOWLEDGMENTS

The first author gratefully acknowledges the support and patience of Professor Dr. Gustaaf Jacobs. Funding by AFOSR (Computational Math and Energetic Material programs) is gratefully acknowledged by Prof. Gustaaf Jacobs. Mr. Reddy Lingampally, Mr. Wayne, and Prof. Vorobieff acknowledge partial support from the US National Nuclear Security

Administration (NNSA). This work is supported by the US National Science Foundation (NSF), CBET Division. Nobody ever says anything nice about Prof. Vorobieff's patience, but that's okay.

REFERENCES

- [1] Rayleigh, L., Analytic solutions of the Rayleigh equation for linear density profiles. *Proceedings of the London Mathematical Society*, **14**, pp. 170–177, 1883.
- [2] Taylor, G., The instability of liquid surfaces when accelerated in a direction perpendicular to their planes. I. *Proceedings of the Royal Society of London A: Mathematical, Physical and Engineering Sciences*, **201**(1065),1 pp. 92–196, 1950. https://doi.org/10.1098/rspa.1950.0052
- [3] Richtmyer, R.D., Taylor instability in shock acceleration of compressible fluids. *Communications on Pure and Applied Mathematics*, 13(2), pp. 297–319, 1960. https://doi.org/10.1002/cpa.3160130207
- [4] Markstein, G.H., Flow disturbances induced near a slightly wavy contact surface, or flame front, traversed by a shock wave, 1957.
- [5] Meshkov, E.E., Instability of shock-accelerated interface between two media. *Advances in Compressible Turbulent Mixing*, 8810234:**473**, 1992.
- [6] Balakrishnan, K., Ukai, S. & Menon, S., Clustering and combustion of dilute aluminum particle clouds in a post-detonation flow field. *Proceedings of the Combustion Institute*, 33(2), pp. 2255–2263, 2011. https://doi.org/10.1016/j.proci.2010.07.064
- [7] Jacobs, G.B. & Don, W-S., A high-order WENO-Z finite difference based particlesource-in-cell method for computation of particle-laden flows with shocks. *Journal of Computational Physics*, 228(5), pp. 1365–1379, 2009. https://doi.org/10.1016/j.jcp.2008.10.037
- [8] Vorobieff, P., Anderson, M., Conroy, J., White, R., Truman, C.R. & Kumar, S., Analogues of Rayleigh-Taylor and Richtmyer-Meshkov instabilities in flows with nonuniform particle and droplet seeding. *Computational Methods in Multiphase Flow VI*, A. Mammoli, and C. Brebbia, eds., WIT Press, Ashurst, Southampton, UK, pp. 17–28, 2011.
- [9] Vorobieff, P., Anderson, M., Conroy, J., White, R., Truman, C.R. & Kumar, S., Vortex formation in a shock-accelerated gas induced by particle seeding. *Physical Review Letters*, **106**(18), p. 184503, 2011. https://doi.org/10.1103/PhysRevLett.106.184503
- [10] Sharma, M.P. & Stock, D.E., The particle-source-in cell (PSI-CELL) model for gasdroplet flows. *Journal of Fluids Engineering*, **99**(2), pp. 325, 1977. https://doi.org/10.1115/1.3448756
- [11] Boiko, V.M., Kiselev, V.P., Kiselev, S.P., Papyrin, A.N., Poplavsky, S.V. & Fomin, V.M., Shock wave interaction with a cloud of particles. *Shock Waves*, 7(5), pp. 275–285, 1997. https://doi.org/10.1007/s001930050082
- [12] Meshkov, E.E., Instability of the interface of two gases accelerated by a shock wave. Soviet Fluid Dynamics, 4(5), pp. 101–104, 1969. https://doi.org/10.1007/bf01015969

70