



## Research on the effect of rotation and curvature on turbulence model and their application

Xin Jiang<sup>1,2\*</sup>, Lei Zhang<sup>1</sup>

<sup>1</sup>College of Mechanical Science and Engineering, Jilin University, Changchun City, 130025, China

<sup>2</sup>College of Civil Engineering, Jilin Jianzhu University, Changchun City, 130118, China

Email: 660-jx@163.com

### ABSTRACT

This paper analyzes the advantages and defects of the  $k-\omega$  SST turbulence model when adopted for simulation calculation under working conditions of rotational motion and curved walls. On the basis of this model and in consideration of the rotational motion and the curvature effect, the author introduces the correction equation proposed by Spalart on the basis of the steady-state rotation effect, and Hellsten's Richardson number correction model, recalculates the specific dissipation rate in the near-wall area by the EWT, and corrects the turbulent kinetic energy generation term in the turbulence model. The author applies the original and the improved turbulence models in the calculation of the lift and efficiency of the centrifugal pump and finds that the calculation results of the improved model are closer to the experimental results at a rotating speed of 2,500n/min or above. The calculation results are optimal under the working condition of a rotating speed of 3,370n/min and a flow of 180L/min. The calculated result of pump lift is 21.04m, with a calculation error of 0.67%, meaning that the calculation accuracy is improved by 0.3%; the calculated result of efficiency is 26.2%, with a calculation error of 0.76%, meaning that the accuracy is improved by 0.7%. After analyzing the inner flow field, the author finds that in the turbulence intensity nephogram calculated by the improved model, the turbulence intensity in the outlet area is low, the forward trend along the curved wall is weak and the velocity is significantly reduced. From the turbulence kinetic energy nephogram, it can be seen that the calculation results of the improved model are low at the outlet and the central area, and that the turbulence development is weak. By comparing the calculation results and experimental results of the original model and the improved one, it can be proved that the improved turbulence model has higher calculation accuracy.

**Keywords:** Rotation and Curvature Effect, Near-Wall Area, Turbulence Model, Centrifugal Pump.

### 1. INTRODUCTION

Turbulence is an important feature of the internal flow field of the centrifugal pump. Due to the special structure of the centrifugal pump and the combined action of Coriolis force and centrifugal force, the structure of turbulence is particularly complicated in the flow channel of the centrifugal pump.

The internal boundary layer of the impeller is greatly impacted by the rotation of the flow field and the curvature of the structure. If the impeller blades are highly curved and the pump is rotating at a high speed, unstable flow phenomena are likely to occur in the flow field, such as separated flow and rotating flow, resulting in internal flow field loss and reduced pump efficiency [1].

Computational fluid dynamics (CFD) provides an effective way to study and design the structure of the internal flow field of the centrifugal pump. Despite the significant improvement on research efficiency by CDF, the calculation accuracy is still affected by the selection and application of the turbulence model [2].

Recent years have seen the popularity of the  $k-\varepsilon$  model and the  $k-\omega$  model, both of which are two-equation models, among engineers and scientific researchers. It is demonstrated by repeated calculations that: although the  $k-\varepsilon$  model can make desirable simulations of some of the complex flows, it fails to reflect the anisotropic properties of the turbulence in the case of a curved flow line because the turbulence viscosity becomes isotropic based on the Boussinesq hypothesis; What is worse, it cannot be applied directly to simulate the near-wall flow which has a small Reynolds number, unless it is improved by the near-wall function [3-4].

The  $k-\omega$  model is developed on the basis of the  $k-\varepsilon$  model. The new two-equation model effectively improves the processing method of the flow of low Reynolds number fluid. Three kinds of  $k-\omega$  models are available currently, namely the Wilcox  $k-\omega$ , the Baseline  $k-\omega$  and the  $k-\omega$  SST. Among them, the  $k-\omega$  SST is the most widely used model [5]. It separates the near-wall area and the turbulent core outside the boundary layer by the hybrid function, and makes effective prediction of the starting point of separation and the separation areas of the fluid under adverse pressure gradient. However, neither

the  $k-\varepsilon$  model or the  $k-\omega$  model support the accurate numerical calculation of the turbulent core and the near-wall area under the flow mode of strong rotation and large curvature.

On the basis of the  $k-\omega$  SST turbulence model and in view of the near-wall function, this paper considers the working conditions of rotating flow field and curved structure, corrects the coefficient of the turbulent kinetic energy generation term of the model by the Spalart rotation curvature correction equation, revises the dissipation term of the  $k-\omega$  equation by Hellsten's Richardson number correction model, and recalculates the dissipation rate in the near-wall area by the EWT-  $k-\omega$  mode. The resulting improved  $k-\omega$  SST model is applied to the calculation of the flow field in the centrifugal pump on the automobile engine, and the effect of the model is verified by the product experiment. Thus, the research of this paper opens up a new trail for flow field calculation under strong rotation and large curvature.

The revised model in this paper is applicable to the flow field simulation calculation of the rotational motion of fluid in a flow channel formed by curved walls at a high rotating speed and mainly addresses the calculation accuracy of the centrifugal pump flow field at a high rotating speed.

## 2. MODEL CORRECTION CALCULATION

### 2.1 Effect of flow mode on turbulence generation term

On the basis of the Reynolds stress equation and the generation rate equation of turbulent energy component, Johnston probed deep into the effect of rotation and curvature on the turbulence structure, and concluded that the rotation and curvature have an impact on the Reynolds stress and its component, and in turn affect the turbulence generation term.

As the Reynolds stress increases in the flow field, the generation rate and level of turbulent energy will rise while the suction face will fall across the board. In terms of the impeller structure, the convex wall suppresses the curvature, the Reynolds stress and turbulent energy level; the concave wall has completely the opposite effect [6-7].

In the transport equation of the SST turbulence model, the generation term is a function based on the strain rate tensor, which does not include the eddy viscosity tensor. In the flow separation area, the strain rate tensor increases rapidly, exerting a significant impact on the Reynolds stress magnitude and turbulent energy level estimated by the model. The flow field has a different mode of motion. In view of the effect of rotation and curvature on the flow field, and in consideration of the eddy viscosity tensor and the strain rate tensor, Spalart put forward the correction function for the turbulence generation term based on the gyroscopic steady-state effect (GSE) [8].

$$f_r = \max \left[ 0, 1 + C_{sca} \left( \tilde{f}_r - 1 \right) \right] \quad (1)$$

$$\tilde{f}_r = \max \left\{ \min \left( f_{rot}, 1.25 \right), 0 \right\} \quad (2)$$

$$f_{r1} \left( r^*, \tilde{f} \right) = \left( 1 + c_{r1} \right) \frac{2r^*}{1 + r^*} \times \left[ 1 - c_{r3} \tan^{-1} \left( c_{r2} \tilde{f} \right) \right] - c_{r1} \quad (3)$$

where  $C_{sca}$  is the scale factor, normally set as 1.0.

$$r^* = S / \omega \quad (4)$$

$$\tilde{f} = 2\omega_{ik} S_{jk} \left( \frac{DS_{ij}}{Dt} + (\varepsilon_{imn} S_{jn} + \varepsilon_{jmn} S_{in}) \omega_m \right) / D^4 \quad (5)$$

$$S_{ij} = 0.5 \left( \frac{\partial u_i}{\partial x_j} + \frac{\partial u_j}{\partial x_i} \right) \quad (6)$$

$$\omega_{ij} = 0.5 \left( \left( \frac{\partial u_i}{\partial x_j} - \frac{\partial u_j}{\partial x_i} \right) + 2\varepsilon_{mji} \omega_m \right) \quad (7)$$

$$S^2 = 2S_{ij} S_{ij} \quad (8)$$

$$\omega^2 = 2\omega_{ij} \omega_{ij} \quad (9)$$

$$D^2 = 0.5 \left( S^2 + \omega^2 \right) \quad (10)$$

where  $DS_{ij}/Dt$  is the Lagrangian differential form of the strain rate tensor;  $\omega_m$  is the rotational angular velocity;  $m$  is valued at 1, 2 and 3, representing the x-direction, y-direction and z-direction respectively; the empirical coefficients cr1, cr2 and cr3 are 1.0, 2.0 and 1.0 respectively.

In the function, is limited to the range of 0-1.25 to avoid excessive turbulence viscosity.  $f_i=0$  means convex curvature flow and turbulence-free steady-state flow, while  $f_i=1.25$  means strong concave curvature flow or high-speed rotation. The upper limit of 1.25 has been proved to be appropriate by a number of case studies [9].

### 2.2 The effect of the mode of motion of the flow field on the boundary layer

The rotating flow field and the curved wall surface have a great influence on the development of the boundary layer. A big calculation error will occur if the boundary layer is treated as a stationary plate in the flow field calculation.

It is found in the analysis of the rotating flow field that the Coriolis force has an effect on the stability of the boundary layer in the rotating state. If the Coriolis acceleration has a component perpendicular to the solid wall, the Coriolis force will enhance or weaken the stability of the turbulence. If the Coriolis acceleration has a component parallel to the solid wall, the Coriolis force will cause a secondary flow perpendicular to the flow direction [6].

Through the observation of the turbulent flow in the 2D rotating groove, Bradshaw obtained the gradient Richardson's number (Ri), and used the parameter to describe the stability of the boundary layer. The parameter is of great importance because it illustrates the effect of rotation and flow line curvature on the flow of the shear layer [10].

$$Ri = -2\omega^{Rot} \left( \frac{\partial U}{\partial y} - 2\omega^{Rot} \right) / \left( \frac{\partial U}{\partial y} \right)^2 = S(S+1) \quad (11)$$

$$S = -2\omega^{\text{Rot}} \left( \partial U / \partial y \right)^{-1} \quad (12)$$

where  $\omega^{\text{Rot}}$  is the system's rotational angular velocity;  $U$  is the fluid velocity;  $y$  is the direction of the radius of curvature; For the centrifugal impeller, the  $Ri$  of the boundary layer of the suction surface is  $> 0$ , i.e. the flow is stable at this position; the  $Ri$  of the boundary layer of the pressure surface is  $< 0$ , i.e. the flow is unstable at this position.

### 2.3 Calculation of the near-wall area

In the near-wall area, a low Reynolds number is an important feature of fluid. It has been proved in numerous tests that the near-wall area is categorized into three parts: the viscous sublayer, the buffer layer, and the turbulence layer, aka. the logarithmic layer. Most turbulence models employ the wall function method to treat the wall area. Nevertheless, if the Reynolds number of the flow  $Re < 10^5$ , the solution of the transport equation often overlooks the viscous sublayer of the boundary layer, leading to an offset error of boundary layer thickness as high as 25%.

The near-wall function provides an equation related to the specific dissipation rate  $\omega$  for the viscous sublayer of the boundary layer, so that the wall function can be converted to the near-wall equation with a low Reynolds number. The main idea of the near-wall function is to adjust the value of  $\omega$  between the logarithmic equation and the linear equation [11].

In the near-wall function, the momentum flux equation is acquired through speed calculation. The equation is expressed as:

$$F_U = -\rho u_\tau^* u^* \frac{1}{2} \quad (13)$$

where

$$u^* = \sqrt{\left( \sqrt{\frac{\mu}{\rho}} \left| \frac{\Delta U}{\Delta y} \right| \right)^4 + (\sqrt{a_1 k})^4} \quad (14)$$

$$u_\tau = \sqrt{(u_\tau^{\text{vis}})^4 + (u_\tau^{\text{log}})^4} \quad (15)$$

$$u_\tau^{\text{vis}} = \sqrt{\frac{\mu}{\rho} \left| \frac{\Delta U}{\Delta y} \right|} \quad (16)$$

$$u_\tau^{\text{log}} = \frac{U}{1 / \kappa \log(y^+) + C} \quad (17)$$

The near-wall function provides an equation related to  $\omega$ , where the  $\omega_l$  of the logarithmic layer of the boundary layer is expressed as:

$$\omega_l = \frac{u^*}{a_1 \kappa y} = \frac{1}{a_1 \kappa \nu} \frac{u^{*2}}{y^+} \quad (18)$$

The  $\omega_s$  of the viscous sublayer of the boundary layer is expressed as:

$$\omega_s = \frac{6\nu}{\beta(\Delta y)^2} \quad (19)$$

where  $\Delta y$  is the distance between the first and second nodes near the wall,  $y^+$  is the dimensionless distance from the centroid of the first layer to the wall. To avoid the oscillatory properties of computational convergence, the following equation is adopted:

$$\omega_\omega = \omega_s \sqrt{1 + \left( \frac{\omega_l}{\omega_s} \right)^2} \quad (20)$$

Since the viscous sublayer of the boundary layer is considered by the  $\omega$  equation in the near-wall function, the corrected wall function can accurately predict the flow of the low Reynolds number boundary layer.

## 3. CORRECTION OF THE TURBULENCE MODEL COEFFICIENT

### 3.1 SST $k$ - $\omega$ turbulence model

The flow field is calculated by the  $k$ - $\omega$  SST turbulence model, which is defined as:

$$\frac{\partial K}{\partial t} + u_j \frac{\partial K}{\partial x_j} = P_K - \beta^* \omega K + \frac{\partial}{\partial x_i} \left[ \left( \nu + \sigma_k \nu_t \right) \frac{\partial K}{\partial x_j} \right] \quad (21)$$

$$\begin{aligned} \frac{\partial \omega}{\partial t} + u_j \frac{\partial \omega}{\partial x_j} &= \gamma \frac{\omega}{K} P_K - \beta \omega^2 \\ &+ \frac{\partial}{\partial x_j} \left[ \left( \nu + \sigma_\omega \nu_t \right) \frac{\partial \omega}{\partial x_j} \right] \\ &+ 2(1 - F_1) \sigma_{\omega 2} \frac{1}{\omega} \frac{\partial K}{\partial x_j} \frac{\partial \omega}{\partial x_j} \end{aligned} \quad (22)$$

where  $K$  is the turbulent kinetic energy;  $P_K$  is the generation term of turbulent kinetic energy;  $P_K = \nu_t S^2 = 2\nu_t S_{ij} S_{ji}$ ;  $u_j (j=1,2,3)$  is the velocity component of the coordinate axis  $x_j$ ;  $\nu$  is the kinematic viscosity coefficient;  $\nu_t$  is the eddy viscosity coefficient;  $\omega$  is the specific dissipation rate;  $\beta^*$ ,  $\sigma_k$ ,  $\gamma$ ,  $\beta$ ,  $\sigma_\omega$ , and  $\sigma_{\omega 2}$  are constants of the transport equation.

The mixed functions:

$$F_1 = \tanh \left( \left( \min \left[ \max \left( \frac{k^{1/2}}{\beta^* \omega y}, \frac{500\nu}{y^2 \omega} \right), \frac{4\sigma_{\omega 2} k}{CD_{k\omega} y^2} \right] \right)^4 \right) \quad (23)$$

$$CD_{k\omega} = \frac{2\sigma_{\omega 2}}{\omega} \frac{\partial k}{\partial x_j} \frac{\partial \omega}{\partial x_j} \quad (24)$$

The eddy viscosity coefficient:

$$\nu_t = \frac{a_1 k}{\max(a_1 \omega, SF_2)} \quad (25)$$

$$F_2 = \tanh \left( \left( \max \left( 2 \frac{k^{1/2}}{\beta^* \omega y}, \frac{500\nu}{y^2 \omega} \right) \right)^2 \right) \quad (26)$$

$$S = (S_{ij} S_{ij})^{1/2} \quad (27)$$

where  $F_1$  and  $F_2$  are the mixed functions used to determine which of the  $k-\omega$  model and the  $k-\epsilon$  model should be used for the calculation area.

### 3.2 Model coefficient adjustment

According to the rotation correction function proposed by Spalart, the turbulence energy generation term  $P_K$  of the turbulence model is corrected [12-13].

$$P_K \rightarrow P_K f \quad (28)$$

The dissipation term is corrected by the Hellsten's Richardson number correction model [14-15], and the correction coefficients for rotation and curvature are defined as follows:

$$F = \frac{1}{1 + C_{rc} Ri} \quad (29)$$

$$Ri = \frac{|\omega_{ij}|}{|S_{ij}|} \left( \frac{|\omega_{ij}|}{|S_{ij}|} - 1 \right) \quad (30)$$

where  $C_{rc} = 3.6$ , and  $\omega_{ij}$  and  $S_{ij}$  are respectively the average rotational tensor and stress rate tensor of the motion, which are calculated by Formulas (6) and (7).

In light of Formulas (28) & (29), the corrected model is expressed as follows:

$$\begin{aligned} \frac{\partial K}{\partial t} + u_j \frac{\partial K}{\partial x_j} &= P_K f_r - \beta^* \omega K \\ + \frac{\partial}{\partial x_i} \left[ (v + \sigma_k \nu_t) \frac{\partial K}{\partial x_i} \right] \end{aligned} \quad (31)$$

$$\begin{aligned} \frac{\partial \omega}{\partial t} + u_j \frac{\partial \omega}{\partial x_j} &= \gamma \frac{\omega}{K} P_K f_r - F \beta \omega^2 \\ + \frac{\partial}{\partial x_j} \left[ (v + \sigma_\omega \nu_t) \frac{\partial \omega}{\partial x_j} \right] - 2(F_1 - 1) \sigma_{\omega 2} \frac{1}{\omega} \frac{\partial K}{\partial x_j} \frac{\partial \omega}{\partial x_j} \end{aligned} \quad (32)$$

where  $\beta^* = 0.09$ ,  $\sigma_k = 2$ ,  $\gamma = 5/9$ ,  $\beta = 0.075$ ,  $\sigma_\omega = 2$ , and  $\sigma_{\omega 2} = 0.44$ . The  $\omega$  of the boundary layer is calculated in combination with Formulas (18), (19) & (20),.

## 4. CALCULATION AND ANALYSIS

### 4.1 Data of case study

This paper applies the coefficient-corrected  $k-\omega$  SST turbulence model to the flow field calculation of the

centrifugal pump of automobile motor. The structural parameters are as follows: the blade  $z=7$ , the impeller diameter is 18mm, the volute is involute, the minimum gap between the shell of volute tongue and the impeller is 0.5mm, and the minimum distance between the bearing surface and the impeller is 1mm. For better solution convergence, a triangular prism mesh network is pulled out from the outlet of water pump with long cooling circulation. The mesh is shown in Figure 1.

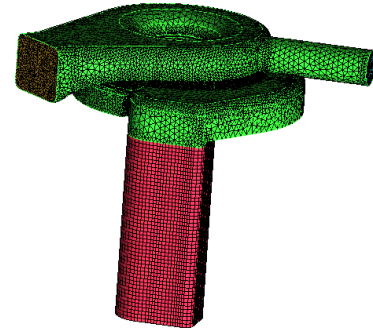


Figure 1. The mesh chart of the centrifugal pump

### 4.2 Calculation implementation and boundary conditions

The coefficients of the model's computational items are adjusted through dynamic loading of the solver by the UDF (User Defined Function) in the fluent software. The adjusted model is defined as  $k-\omega$  SST—RCE.

The boundary conditions are set as follows:

The flow of coolant in the pump is regarded as a steady-state, adiabatic, incompressible turbulence; the inner wall of the pump is considered to be hydraulically smooth; the uncoupled implicit algorithm is adopted, with the second order of accuracy; the coolant's physical parameters are selected according to the test; the mass flow inlet and outflow outlet are chosen; the information transfer between the dynamic and static parts are treated by the GGI.

### 4.3 Calculation results and analysis

This paper applies the  $k-\omega$  SST model (model 1) and  $k-\omega$  SST—RCE model (model 2) to the steady-state turbulence computation of the centrifugal pump of the automobile motor under different working conditions. To avoid the interference from the mesh, the author designs 5 mesh division plans (Table 1). As shown in the table, the pump lift results of the two model only have slight changes when the mesh number is greater than 1.77 million. Thus, Plan 3 is adopted by this paper for follow-up study.

#### 4.3.1 Characteristic data analysis

Table 2 displays the calculated results of characteristic data of the centrifugal pump at different rotating speeds and at different flows with a fixed speed, and lists the data of the centrifugal pump obtained in the test by the manufacturer.

Figure 2 to Figure 7 display the curves of the pump lift and efficiency calculated by the  $k-\omega$  SST model and the  $k-\omega$  SST—RCE model at different rotating speeds and at the design flow of  $Q$ ,  $0.8Q$  and  $0.6Q$ , as well as the curves of the pump lift and efficiency measured in the test by the manufacturer.

It can be seen from the curves in Figures 2 to 5 that the results of both models stay close to the test curves, and the accuracy errors are within 5%. Comparatively speaking, the results of the  $k-\omega$  SST—RCE model are closer to the test values. In the working condition of a rotating speed of 3,370n/min and a flow of  $0.6Q$ , the results of the  $k-\omega$  SST—RCE model are very similar to the test results. The calculated pump lift is 0.3% more accurate than that of the  $k-\omega$  SST model, the calculated efficiency is about 0.7% more accurate than that of the latter.

**Table 1.** The mesh division plan and the calculation results

Plan	Mesh number ( $10^4$ )	The pump lift calculated by model 1 H/m	The pump lift calculated by model 2 H/m
1	136	16.11	16.65
2	154	18.09	17.20
3	177	18.12	18.03
4	199	18.18	18.06
5	221	18.17	18.05

**Table 2.** The experimental data and model calculation results

Gro up	N (n/min)	Q (L/min)	Measured pump lift (H/m)	H <sub>1</sub>	H <sub>2</sub>	$\Delta H_1$ (H <sub>1</sub> -H)	$\Delta H_2$ (H <sub>2</sub> -H)	$\Delta H_1/H(\%)$	$\Delta H_2/H(\%)$
1	3370	300	17.875	18.12	18.03	0.245	0.155	1.35	0.86
2	3370	240	19.81	20.13	20.05	0.32	0.24	1.59	1.20
3	3370	180	20.9	21.09	21.04	0.19	0.14	0.90	0.67
4	2500	240	10.56	11.08	11.02	0.52	0.46	4.69	4.17
5	2500	192	11.01	11.51	11.45	0.5	0.44	4.34	3.84
6	2500	144	12.19	12.6	12.47	0.41	0.28	3.25	2.25
7	2000	150	7.75	6.32	6.43	-1.43	-1.32	-22.63	-20.53
8	2000	120	8.03	6.96	6.87	-1.07	-1.16	-15.37	-16.89
9	2000	90	8.18	7.31	7.14	-0.87	-1.04	-11.90	-14.57
Gro up	N (n/min)	Q (L/min)	Measured efficiency ( $\eta/\%$ )	$\eta_1$	$\eta_2$	$\Delta \eta_1$ ( $\eta_1-\eta$ )	$\Delta \eta_2$ ( $\eta_2-\eta$ )	$\Delta \eta_1/\eta(\%)$	$\Delta \eta_2/\eta(\%)$
1	3370	300	31	32.3	31.8	1.3	0.8	4.02	2.52
2	3370	240	28.4	29.3	28.9	0.9	0.5	3.07	1.73
3	3370	180	26	26.4	26.2	0.4	0.2	1.52	0.76
4	2500	240	35.3	36.7	36.2	1.4	0.9	3.81	2.49
5	2500	192	32.1	33.2	32.8	1.1	0.7	3.31	2.13
6	2500	144	26.8	27.7	27.5	0.9	0.7	3.25	2.55
7	2000	150	35.5	31.9	33.8	-3.6	-1.7	-11.29	-5.03
8	2000	120	31.1	29.8	28.6	-1.3	-2.5	-4.36	-8.74
9	2000	90	25.6	24.7	23.9	-0.9	-1.7	-3.64	-7.11

In Figures 6 and 7, the curves of the numerical results of the two models intersect each other. If the rotating speed remains at a low level, the numerical results of the  $k-\omega$  SST—RCE are higher with a big flow, and lower with a small flow. Through comparison with the measured data, it is found that the calculated results of pump lift under different flow conditions all have errors of more than 10%, the biggest of which is up to 22.63%. The calculated results of efficiency under different flow conditions all have errors of more than 3%, the biggest of which is up to 11.29%. The errors are too large, and thus it is deemed that the results are not correct. In the causal analysis, in order to compare the calculation

accuracy of different models, the same condition is set up for all working conditions – the flow field is steady on the interface between the rotating and stationary areas. However, when the pump rotates at a low speed, the rotating speed of the flow field in the pump decreases. The flow in the flow field is transient and pulsatile, and the flow on the interface between the rotating and stationary areas can no longer be regarded as steady and inhomogeneous, that is, it no longer satisfies the prerequisite for simulation with the MRF (Multiple reference frame) model, leading to large errors in the calculated results.

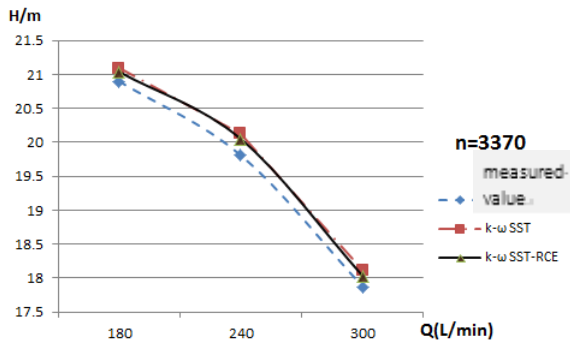


Figure 2. The pump lift curves at  $n=3,370/Q=300$

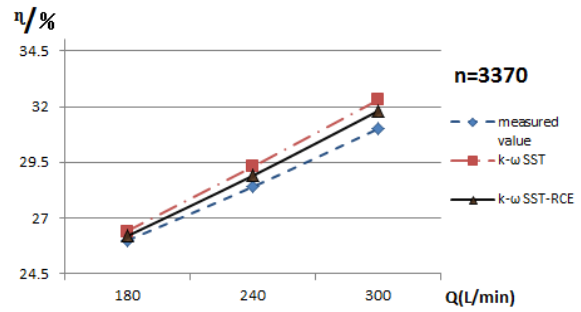


Figure 3. The efficiency curves at  $n=3,370/Q=300$

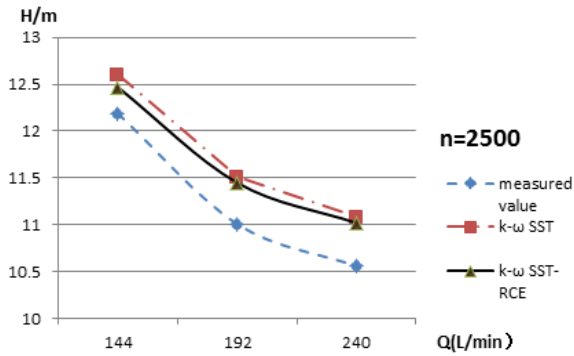


Figure 4. The pump lift curves at  $n=2,500/Q=240$

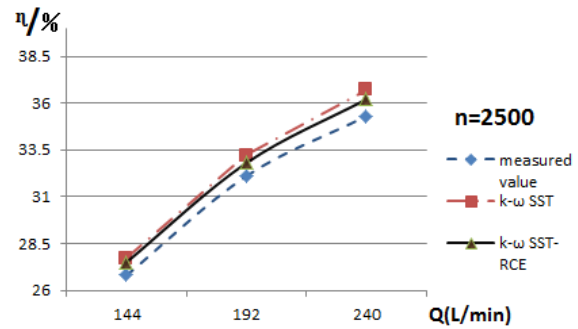


Figure 5. The efficiency curves at  $n=2,500/Q=240$

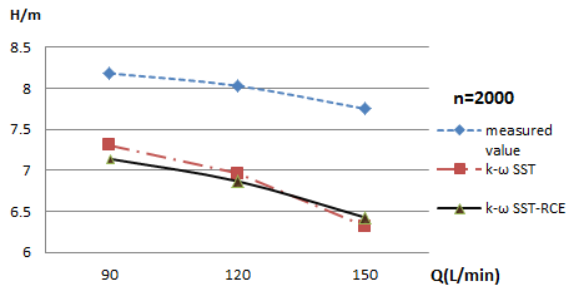


Figure 6. The pump lift curves at  $n=200/Q=150$

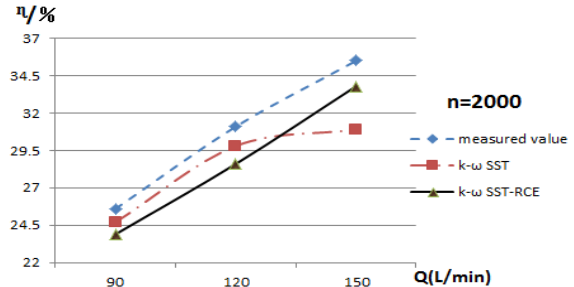


Figure 7. The efficiency curves at  $n=200/Q=150$

#### 4.3.2 Analysis of flow field results

The characteristic data analysis points out that, at a high rotating speed, the  $k-\omega$  SST—RCE model (model 2) offers a better illustration of the motion of a fast rotating and highly curved flow field than the  $k-\omega$  SST model (model 1). To make a comprehensive comparison between the characteristics and differences of the two models, the author chose to analyze the following three aspects of the flow field calculation at a rotating speed of 3,370n/min and a flow of 180L/min.

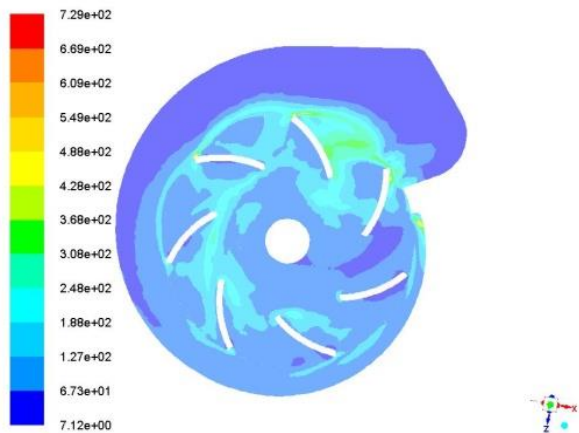
Turbulence intensity and viscosity analysis. Figures 8-11 shows the turbulence intensity and viscosity nephograms of the flow field. The trends of the two parameters are largely the same.

When it comes to the turbulence intensity at the outlet expansion area, however, the calculated results of the  $k-\omega$  SST—RCE model are apparently lower than that of the  $k-\omega$  SST model. That is because the former model takes into account the effect of rotation on the turbulence structure. Under the action of the centrifugal force, the near-wall fluid

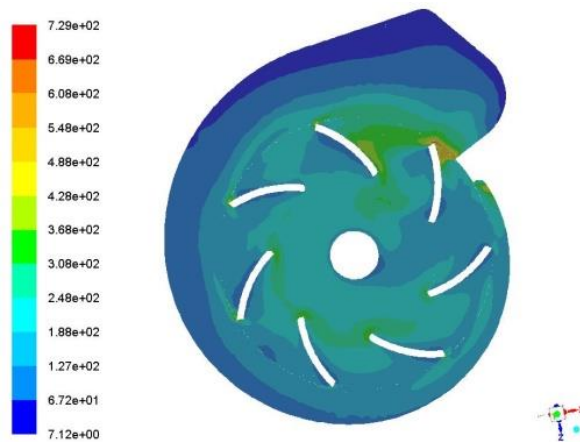
has to overcome the adverse pressure gradient. But the turbulence intensity becomes weaker and the area grows larger. In particular, the turbulence intensity gradually declines at the outlet.

On the suction surface, the fluid flow is not stable under the rotational effect. The turbulence is more intense at the contact part at the tip of the blade due to the blade structure and the mode of fluid motion. As for the central area, model 2 has lower results than model 1, for the turbulence intensity of model 2 is lower in the central area. In this area, the fluid accelerates towards the surrounding areas. The turbulence formation trend is weak before the fluid reaches the tip of the blade. The trend is also demonstrated in the blade speed nephograms in Figure 12 and Figure 13.

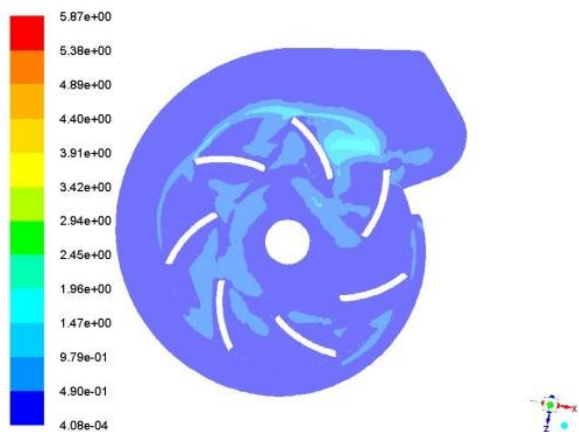
In contrast, in model 1, i.e. the  $k-\omega$  SST model, the turbulence intensity in the expansion area follows the same trend within the blade integrated disk. Besides, model 1 does not consider the effects of rotation and centrifugal force. The above factors contribute to the stronger turbulence intensity and higher calculated results of this model.



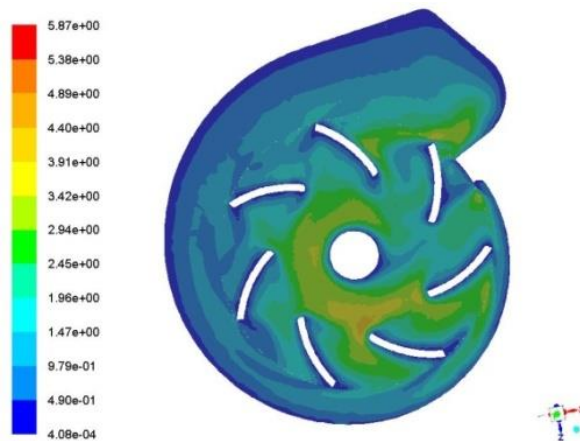
**Figure 8.** The turbulence intensity of the  $k-\omega$  SST



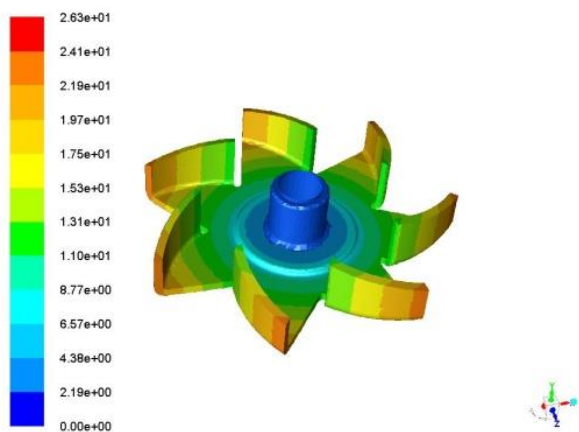
**Figure 9.** The turbulence intensity of the  $k-\omega$  SST—RCE



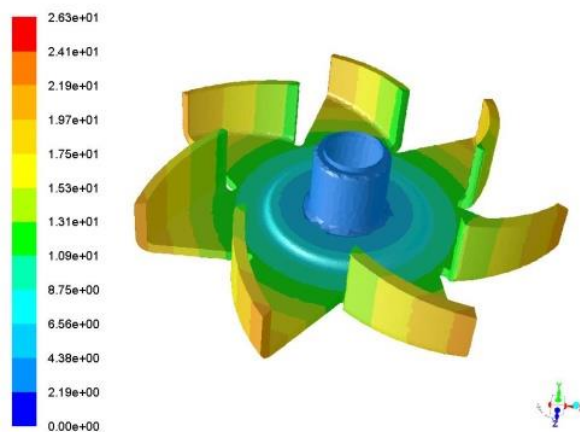
**Figure 10.** The turbulence viscosity of the  $k-\omega$  SST



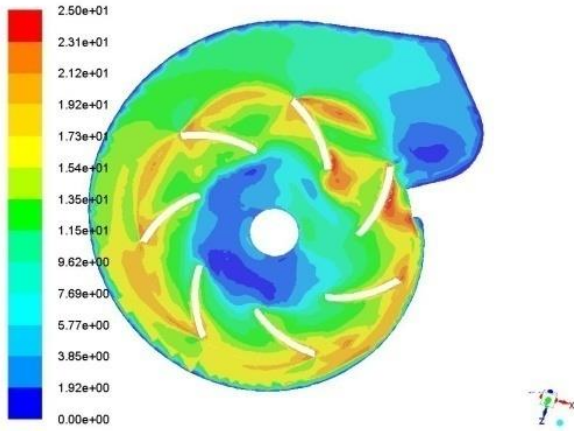
**Figure 11.** The turbulence viscosity of the  $k-\omega$  SST—RCE



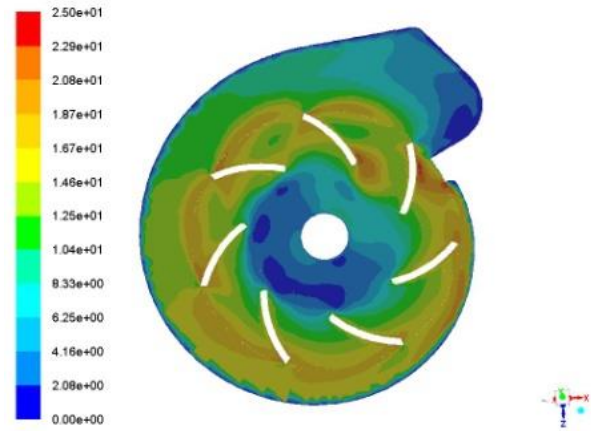
**Figure 12.** The speed of the blade integrated disk of the  $k-\omega$  SST



**Figure 13.** The speed of the blade integrated disk of the  $k-\omega$  SST—RCE



**Figure 14.** The flow field velocity of the  $k-\omega$  SST

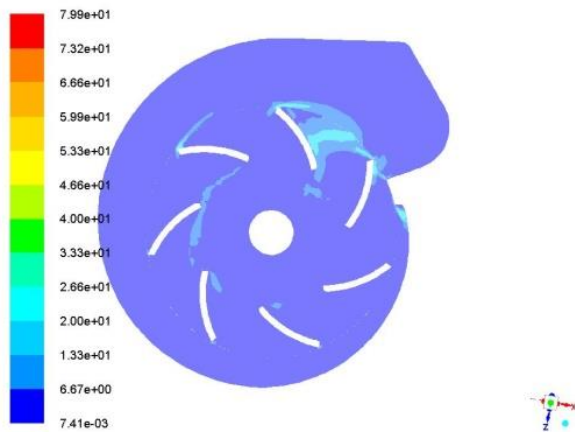


**Figure 15.** The flow field velocity of the  $k-\omega$  SST—RCE

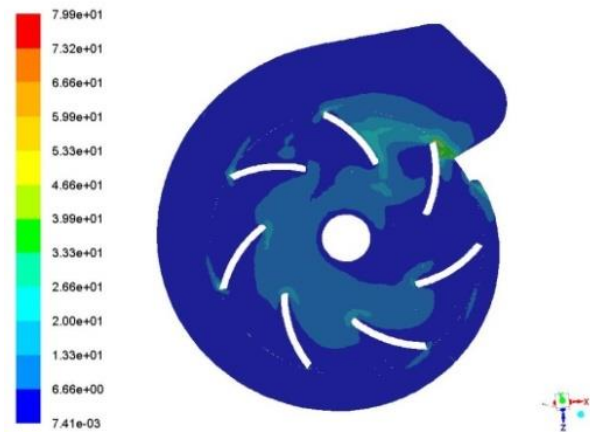
Flow field velocity analysis. The flow field velocity is shown in Figures 14 and 15. The figures display the cross section at 50% of the blade height in the y-direction of the flow field. The outlet velocity is obviously lower in the nephogram of model 2 than in the nephogram of model 1. The fluid motion is not linear at the outlet. Under the action of rotation and curvature, the normal component of the flow field velocity, coupled with the generation of secondary flow, impedes the mixture of low-velocity fluid and the wake flow and expands the area of low-velocity. The effect persists all the way to the outlet expansion section. In the meantime, the fluid has a tendency to move along the volute wall due to volute curvature and rotation. Considering the impact of the wall, the fluid in model 2 tends less to move forward and is more likely to slow down rapidly. Thus, model 2 has a lower outlet velocity, which is consistent with the results of

turbulence intensity and viscosity analysis, as well as the calculation results. In the central area, the low-velocity area in model 2 is larger than that in model 1, and the velocity grows significantly after the fluid passes the tip of the blade.

The effect of turbulent kinetic energy. Turbulent kinetic energy is a measure of the development or decline of turbulence. It is  $\frac{1}{2}$  of the product of the turbulence velocity fluctuation variance and fluid mass. Figures 16 and 17 are the nephograms of the turbulent kinetic energy calculated by model 1 and model 2. As shown in the figures, model 1 boasts a well-developed turbulence, featuring higher turbulent kinetic energy at the outlet and the central area; model 2, however, has lower turbulent kinetic energy in corresponding areas, a signal of low turbulent velocity. The pattern agrees well with the external characteristic calculation. It also indicates that model 2 is good at capturing turbulence.



**Figure 16.** The turbulent kinetic energy of the  $k-\omega$  SST



**Figure 17.** The turbulent kinetic energy of the  $k-\omega$  SST—RCE

## 5. CONCLUSION

In view of the mode of motion of the flow field in the centrifugal pump, this paper analyzes the advantages and drawbacks of the  $k-\omega$  SST turbulence model when it is adopted for simulation calculation under the working condition of rotational motion and curved walls. On the basis of the model and in the aid of near-wall function, the author corrects the coefficients of the turbulent kinetic energy generation term and dissipation term of the model according

to the flow field rotation and curvature effect correction methods proposed by Spalart and Hellsten. The corrected model is applied to the numerical calculation of the centrifugal pump of an automobile motor. In comparison with the original model, the author draws the following conclusions:

1) The calculation results of the external characteristics of a centrifugal pump at a high rotating speed from the improved model are closer to the experimental results. Compared with that of the original model under different flow conditions at a



high rotating speed, the calculation efficiency is improved. In the working condition of a rotating speed of 3,370n/min and a flow of 180L/min, the calculated result of the pump lift is 21.04m, with a calculation error of 0.67%, meaning that the calculation accuracy is improved by 0.3%; the calculated result of efficiency is 26.2%, with a calculation error of 0.76%, meaning that the accuracy is improved by 0.7%.

2) The author compares the calculated results of the two models at 3370n/min and 180L/min, and analyzes the nephograms of the calculation results. In terms of external characteristics, the calculated result of the pump lift in model 2 is smaller than that in model 1, which is closer to the measured value. From the nephograms, it can be seen that the turbulence intensity at the outlet area in model 2 is lower than that in model 1, that the intensity range is greater and that the turbulence intensity at the outlet area lessens along the direction of the fluid motion; in model 2, the speed at the outlet area displayed in the velocity nephogram is also lower than that in model 1. Under the action of rotation and curvature, the normal component of the flow field velocity, coupled with the generation of secondary flow, impedes the mixture of low-velocity fluid and the wake flow and expands the area of low-velocity. The effect persists all the way to the outlet expansion section. The velocity is also gradually decreased along the direction of fluid motion, unlike in model 1, where there is a great velocity gradient. At the same time, the nephograms also show that the fluid at the outlet has a tendency to move along the volute wall, but in model 2, the forward tendency is weak, which is the result of the near-wall treatment of the boundary layer. The turbulence intensity at the outlet is weak, and the flow field velocity is low, so the fluid velocity at the outlet is low, with a low lift. From the above two aspects, it can be seen that the calculated result of pump lift in model 2 is closer to the measured value.

3) When the rotating speed is 2000n/min or lower, the calculated result from the two models are lower than the measured values, and the errors for both pump lift and efficiency are over 3%. In model 1, the maximum error in the calculated result of pump lift is up to 22.36%, thus the result is considered incorrect. The setup of the models under different rotating speeds affects the calculated results. At a low rotating speed, the flow on the interface between the rotating and stationary areas of the pump can no longer be regarded as steady and inhomogeneous, that is, it no longer satisfies the prerequisite for simulation with the MRF (Multiple Reference Frame) model, leading to large errors in the calculated results. As the flow field at a low rotating speed is transient and pulsatile, the calculations regarding the interface should consider the relative sliding of grids and also the interactions between the two areas using a transient calculation method.

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## NOMENCLATURE

$CD_{k\omega}$	positive portion of the cross-diffusion in ω- transport equation
$f_{r1}$	rotation/curvature correction function of the SARC model
$I^*, \tilde{I}$	nondimensional criteria of rotation/curvature effects defined in Sec.
$S_{ij}$	components of the mean strain tensor
$\omega_{ij}$	components of the vorticity tensor
$\omega_m$	components of the system rotation rate vector

$\varepsilon_{mji}$	tensor of Levi–Civita	$y^+$	nondimensional distance from the surface
$c_{r1}, c_{r2}, c_{r3}$	additional empirical constants of the SA rotation and/or streamline curvature (SARC) model	$\beta^*, \gamma, \beta$	turbulence-model coefficients
$Ri$	Richardson number	$\nu$	dynamic viscosity
$C_{rc}$	constant coefficient in the rotation and curvature sensitization	$\nu_t$	turbulent viscosity
$Re$	Reynolds number	$k$	kinetic energy of turbulence
$u_\tau$	friction velocity	$\sigma_k, \sigma_\omega$	turbulence-model coefficients